

# ***Common Range Integrated Instrumentation System (CRIIS)***

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## ***Common Range Integrated Instrumentation System (CRIIS) Informational Briefing***



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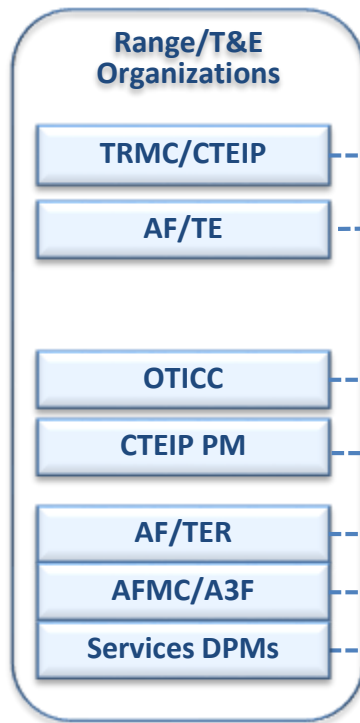
# ***CRIIS Overview***

- **CRIIS is a Family of Systems for Airborne Data Collection**
  - High Accuracy Time, Space, Position Information (TSPI)
  - Secure Datalink(s) Transmit Real Time TSPI and Aircraft Data
  - Multiple Independent Levels of Security
  - Participant Packages (Pod & 2 Internal Configurations) and Ground Systems
  - Integrated on F-15, F-16, F-18, F-22A (SIL) and F-35 (SIL)
- **Development Funded by Central T&E Investment Program (CTEIP)**
  - Replaces Existing GPS Based ARDS at Major Test Ranges – Interoperable System
  - \$300M Development Program
- **Production and Sustainment Funded by Each Service**
- **Prime Ktr: Rockwell Collins, Inc. (Cedar Rapids, IA)**

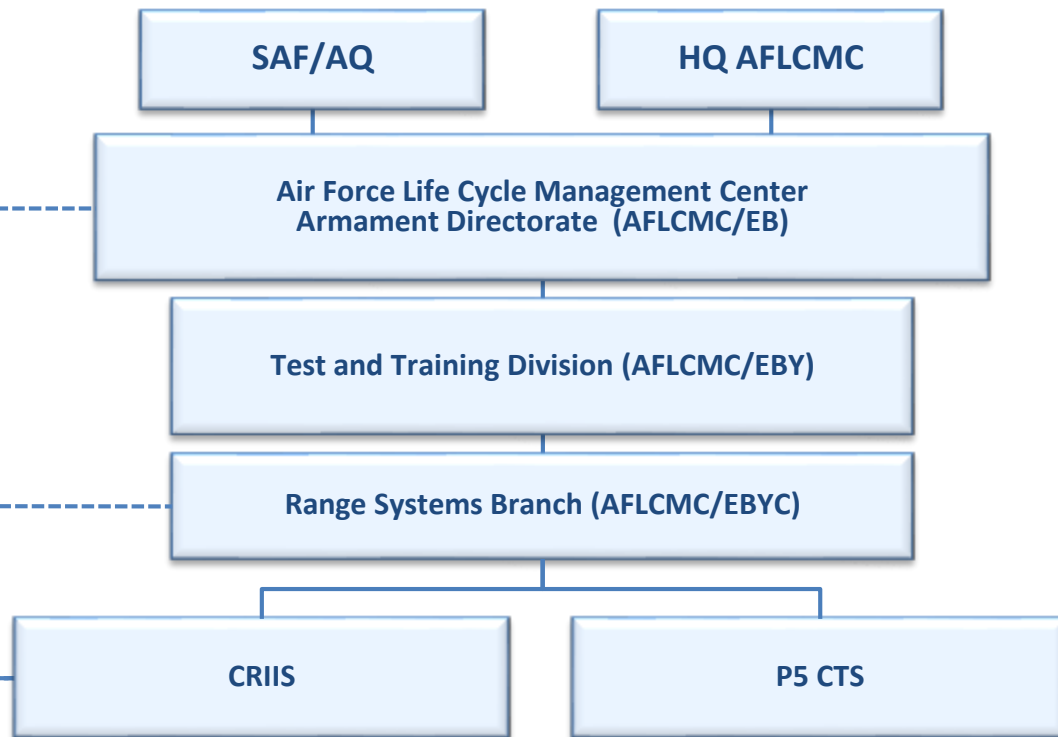
***CRIIS is a Test Range Replacement of the Existing GPS Based ARDS  
with Advanced Datalink, TSPI, Security Features***

# OSD - AFLCMC Relationship

## Requirements/ Funding

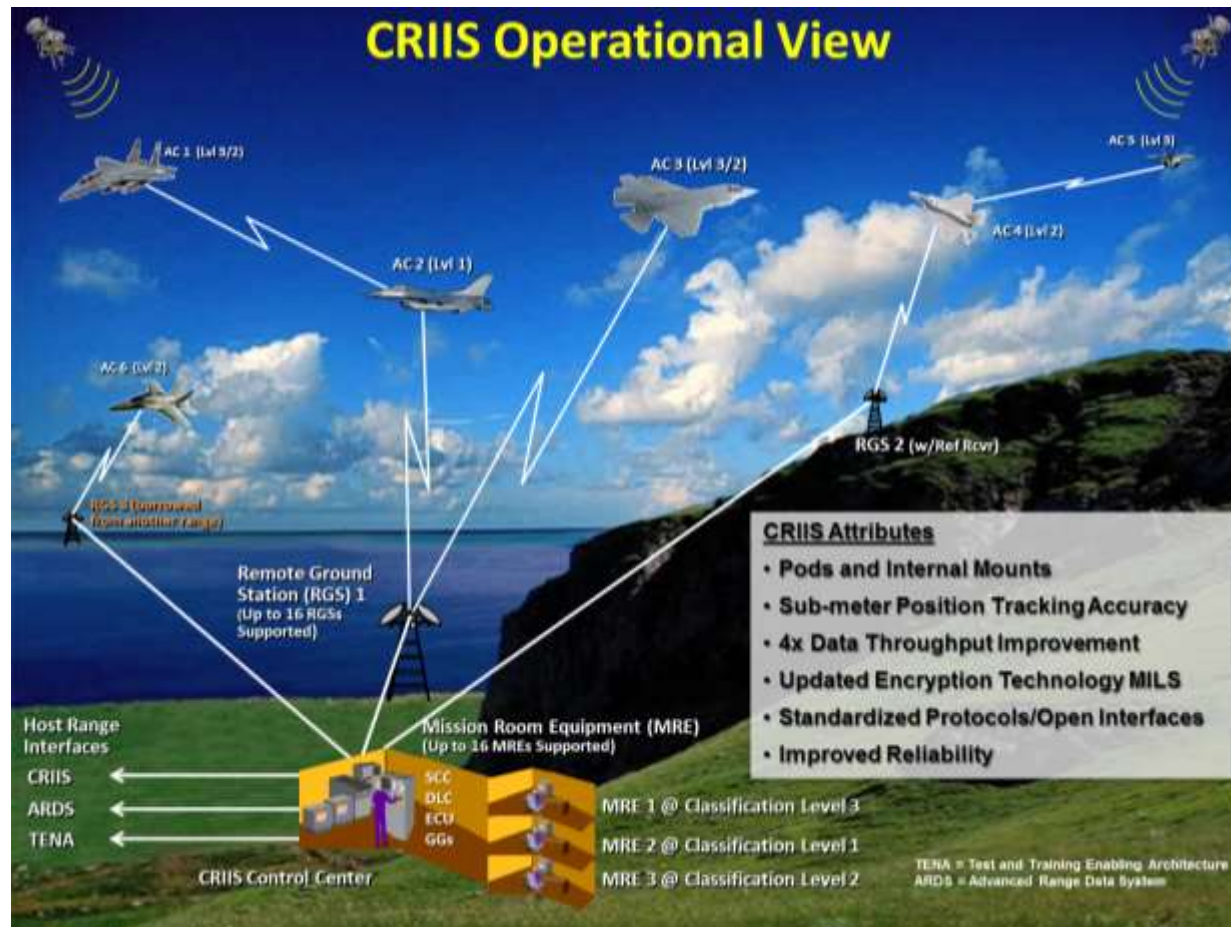


## Acquisition Authority



# CRIIS Description

## CRIIS Operational View



CRIIS is a Family of Systems for Airborne Data Collection

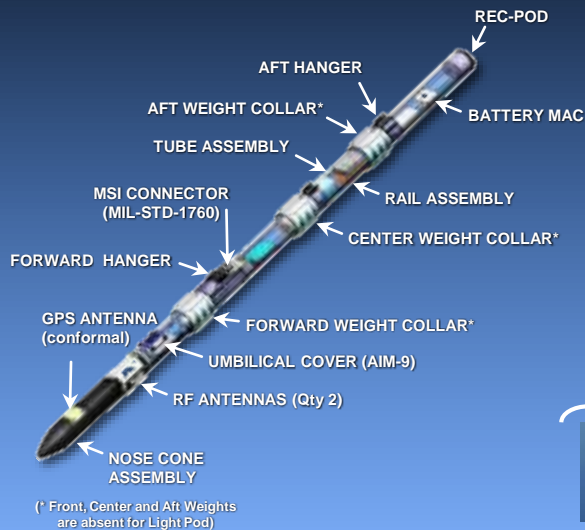
- High Accuracy Time, Space, Position Information (TSPI)
- Increased throughput of Real Time TSPI and Aircraft Data
- Multiple Independent Levels of Security (MILS)
- Participant Packages (Pod & 2 Internal Configurations) & Ground Systems
- Integrating on F-15, F-16, F-18, F-22A (SIL) and F-35 (SIL)
- Replaces Advanced Range Data System (ARDS)

**Phase II locations: Pax River, Eglin AFB, WSMR, and Edwards AFB**  
**Follow-on locations: China Lake, Nellis AFB, and Point Mugu**

# CRIIS Airborne & Ground Subsystems

## Participant Packages

### Configuration 4 Pod for Legacy Aircraft



### Configuration 5 Internal Mount Liquid Cooled for 5<sup>th</sup> Gen Aircraft



### Configuration 6 Internal Mount Air Cooled for Legacy Aircraft



## Threshold Platforms



F-22



F-35



F-15



F-16



F-18

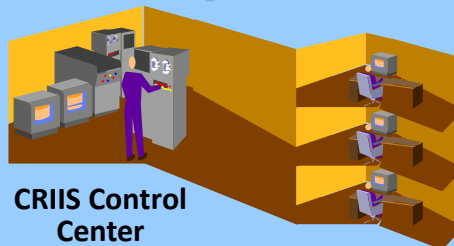
5<sup>th</sup> Generation Aircraft

Legacy Aircraft

## Ground Subsystems



Remote Ground Station & Reference Receiver



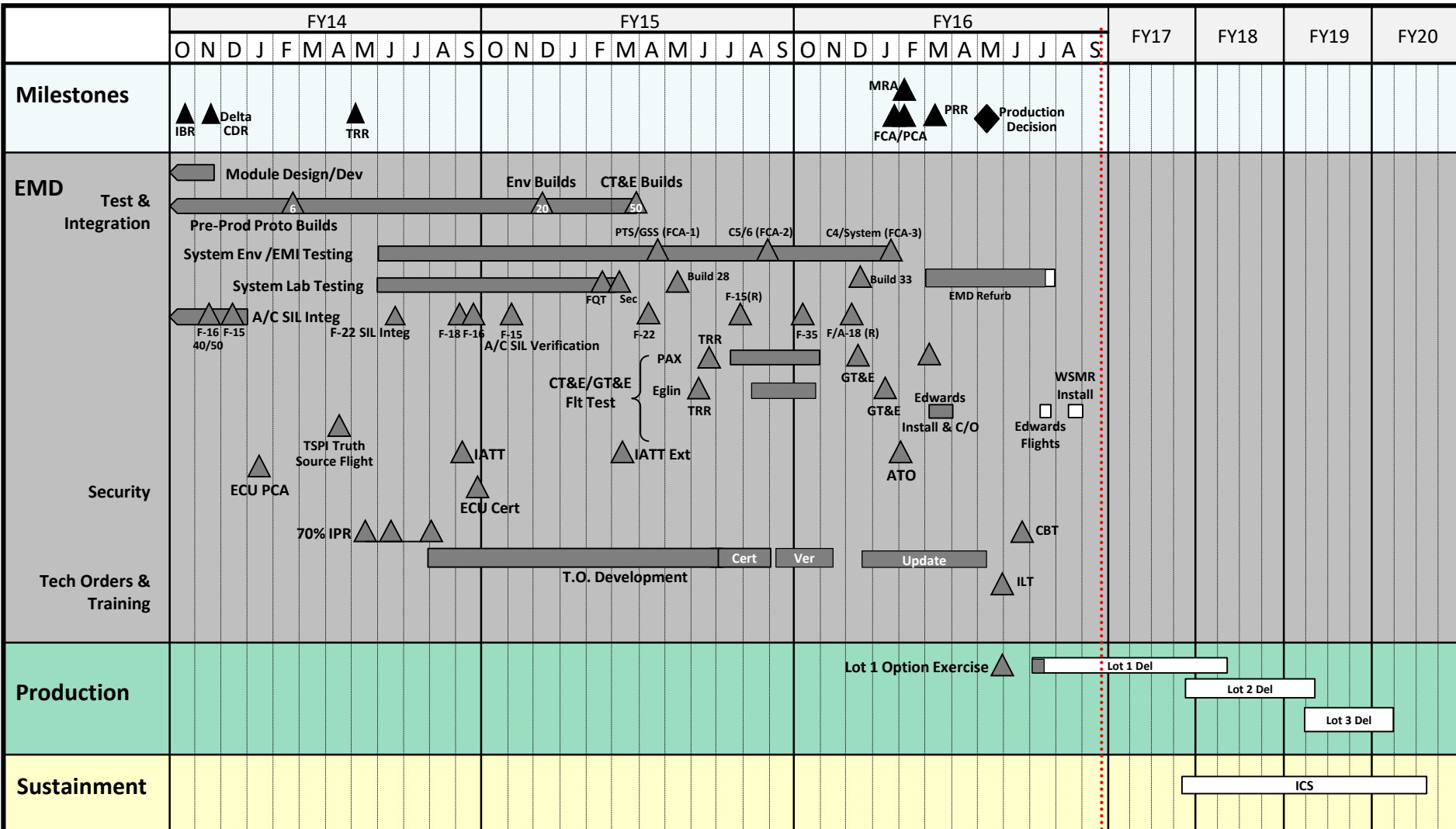
CRIIS Control Center

Mission Rooms

## Portable Test Set

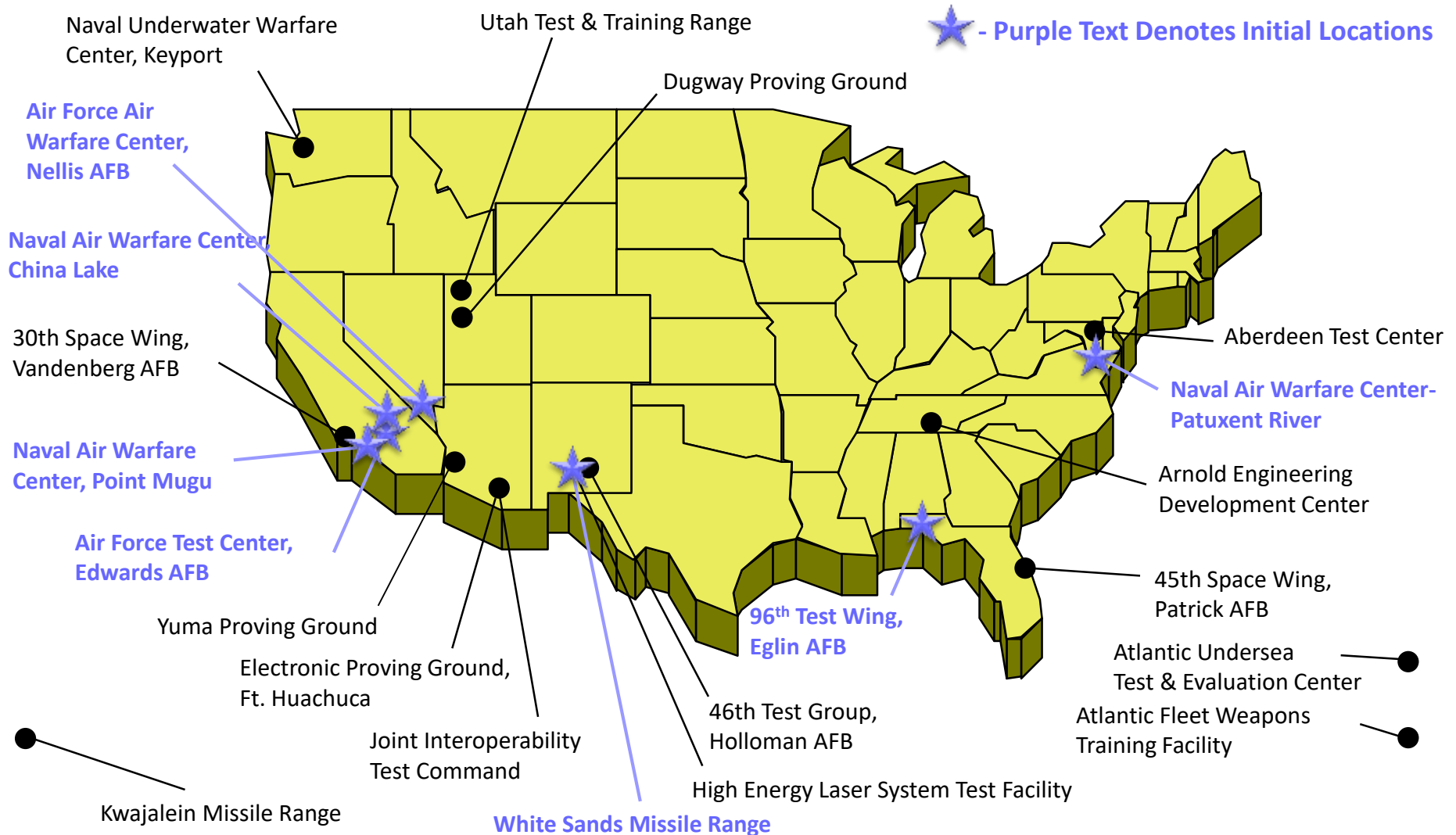


# CRIIS Program Schedule (EMD Detail)





# Major Range and Test Facility Base (MRTFB) and Initial Locations



# Key Performance Parameters

Requirement	Status
0.5 meter horizontal RMS accuracy on fighter aircraft	✓ Met
Top Secret encryption capability	✓ Met
Datalink throughput greater than or equal to 400kbps per frequency within ARDS occupied bandwidth	✓ Met
Mass properties consistent with ARDS, within constraints	✓ Met
Fits internally in F-35 and F-22	See Notes*
Fits on F-18 6L Bay Door	✓ Met
Compatibility with Global Information Grid (through TENA)	✓ Met

\* Notes:

1. Design meets volume requirements defined in SPS at EMD start
2. F-22 aircraft configuration changes (post CRIIS EMD start) will require follow-on integration effort for physical fit.



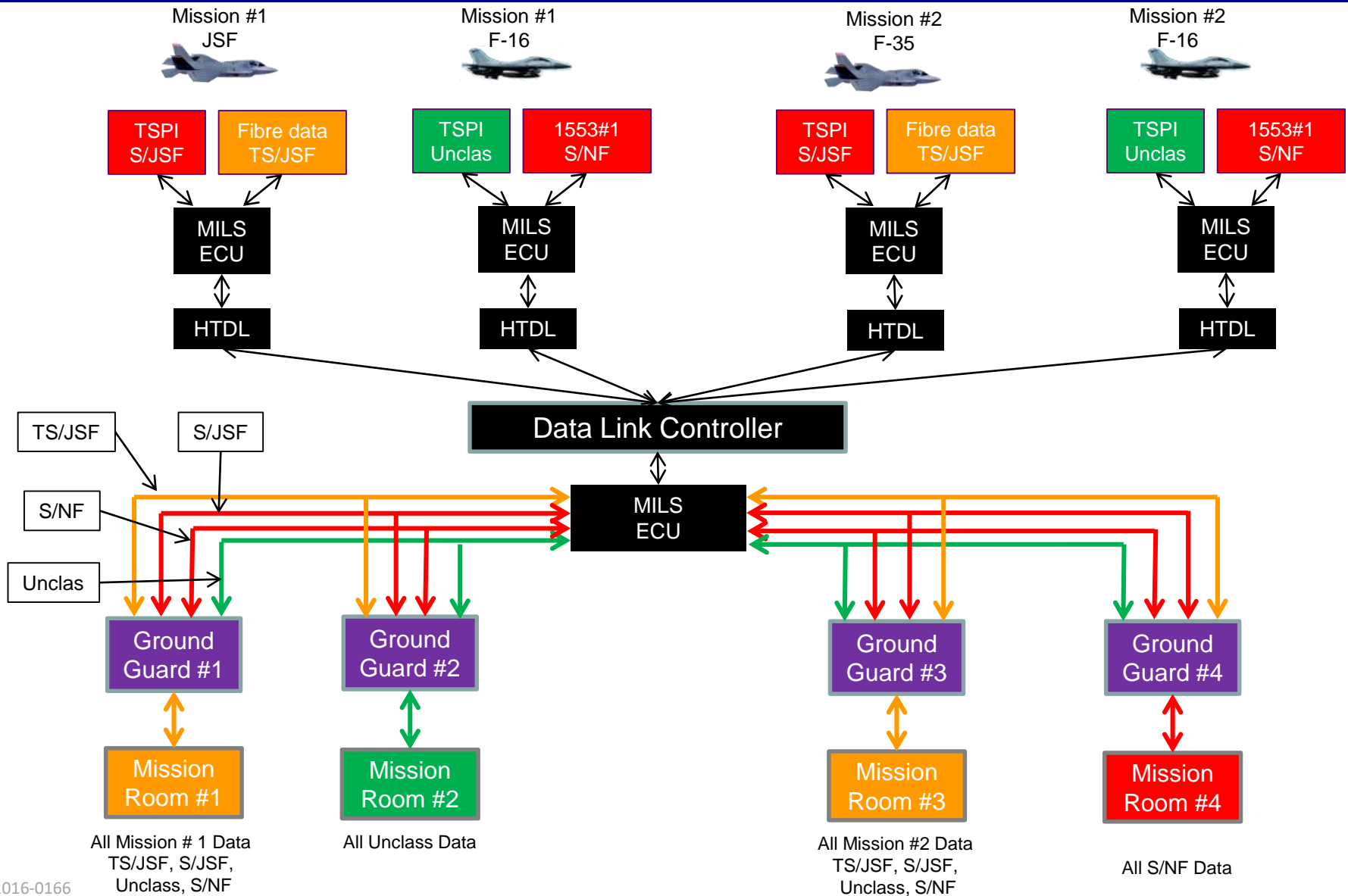
# ***Airworthiness and SIL Testing***

- **SIL Testing Approach**
  - **Demonstrate Aircraft Interface Capability for All Threshold Aircraft**
- **CTE/GTE Flight Clearance Approach**
  - **Obtain Interim/Local Flight Clearances for Flights**
    - **Prioritize Qualification Testing needed for these Clearances**
    - **Subset of Artifacts/Data Needed for Full Fleet Certification**

Aircraft	SIL Objectives
F-15	Cfg4, Cfg6 Power & 1553 Bus Interface
F-16	Cfg4, Cfg6 Power & 1553 Bus Interface
F/A-18	Cfg4, Cfg6 Power & 1553 Bus Interface
F-22	Cfg5 Power & 1553 Bus Interface
F-35	Cfg5 Fiber Channel Bus Interface

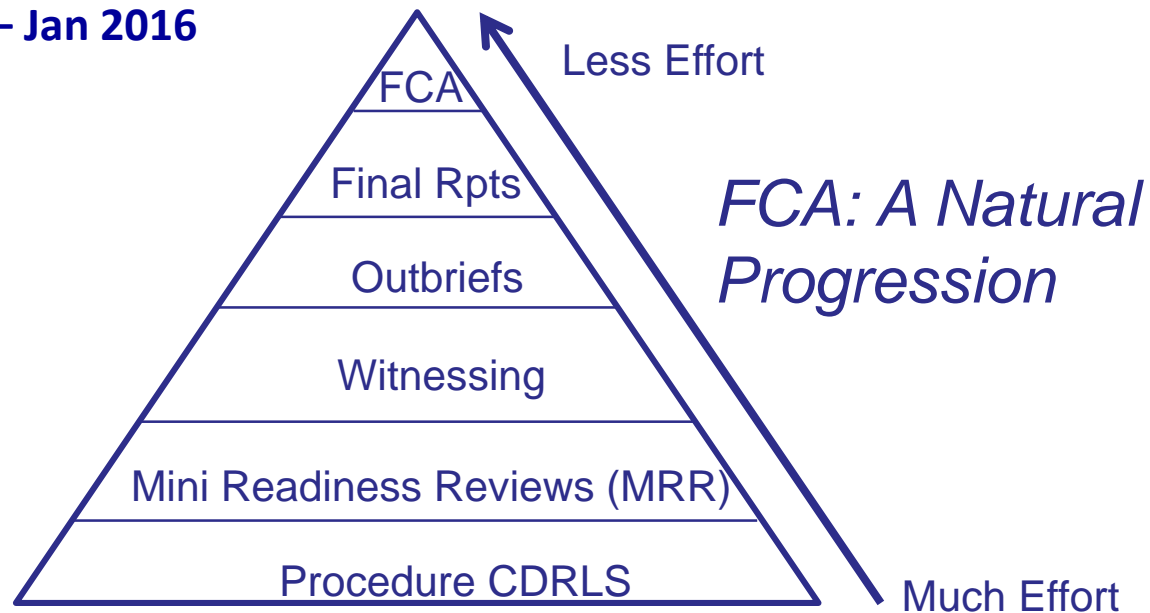
# CRIIS MILS Architecture

*Data Sorted By Classification/Caveat or Mission Number (Notional)*



# Getting to FCA

- Adopted Incremental FCA Approach
  - DoD 5000 accepted practice
  - Volume of Requirements conducive to incremental approach
  - Identifies issues early
- ✓ FCA 1 (GSS/PTS) – Apr 2015
- ✓ FCA 2 (Config 5/6) – Aug 2015
- ✓ FCA 3 (System) – Jan 2016



# *Physical Qual Tests*

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- Testing was performed against multiple product configurations
  - Gen 2: Flight Test Configuration
  - Gen 3: Production Configuration
- Since TRR, over 10,000 hours of Physical Qual Testing
  - EMI, 704, TEMPEST, Dynamics, Climatics
- Physical Qual Testing Complete & Reports Approved
  - Ground: TEMPEST, EMI, ENV
    - CCC, MRE, RGS, RRS, PTS
  - Cfg 4: TEMPEST, EMI, ENV
  - Cfg 5: TEMPEST, EMI, ENV
  - Cfg 6: TEMPEST, EMI, ENV
- Variances Captured

# ***Functional Qual Tests***

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- **Testing was performed against multiple Software Baselines**
  - **Lab FQT: Builds 16, 17, 18, 19, 20, 21, 26**
  - **Lab Regression FQT: Build 28**
  - **Lab Regression FQT: Build 33**
- **Since TRR, over 10,000 hours of Functional Qual Testing**
- **Lab FQT Complete & Reports Approved**
  - **Lab FQT Report**
  - **Build 28 Regression FQT Report**
  - **Build 33 Regression FQT Report**
- **Variances Captured**

# The Process

Rockwell Collins Verification Requirements Traceability Matrix (VRTM)

Requirement (Customer Requirements: Attachment 2-SPS v3 1 thru P41 Located under contract Docs>Attachments)	RC SPS ID	Verif	Associated L2s (Strikethrough indicates no Verif case in DOORS for L2)	Artifact	RC Pass / Fail	Eglin Concurrence	Comments
The HTDL shall detect the presence of message errors in real-time.	SPS-1903	Verif-686 GSS Network Control and RF Interface	<del>L2-1328</del> The Ground Subsystem shall provide the capability to control and monitor all SR-DLTs, MR-DLTs, ER-DLTs, and HT-DLTs simultaneously that are defined as part of the CRIIS system for a specific range.	Lab FQT Rpt 26 (946-9164-001)	Pass	30-Jul-15	HTDL FQT Test Case 09-01
		Verif-2708 HTDL Test	<del>L2-1218</del> The MAC Participant Package shall detect the presence of received message errors in real-time. <del>L2-5016</del> The Ground Subsystem HTDL network shall detect the presence of message errors in real-time.	HTDL FQT Rpt (946-4267-001)	Pass	30-Jul-15	HTDL FQT Test Case 09-01
		Verif-2710 HTDL Analysis	<del>L2-1218</del> The MAC Participant Package shall detect the presence of received message errors in real-time. <del>L2-1826</del> The MAC Participant Package shall have an undetected erroneous message rate of less than one per 16 million. <del>L2-4550</del> The Ground Subsystem HTDL shall have an undetected erroneous message rate of less than one per 16 million.	HTDL FQT Rpt (946-4267-001)	Pass	30-Jul-15	HTDL FQT Figure 3-241

CRIIS Program Office Artifacts Review

4	RC Test Procedure Report Lab Module	RC Verification Cases Verif-686: GSS Network Control and RF Interface Verif-2708: HTDL Test Verif-2710: HTDL Analysis
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Verification Matrix: Click [here](#) for more information on [VTRM](#) traceability

**Issues Review (MRR)**  
[Lab Range MRR](#)

**Issues**  
 Link(s): [Lab Test Procedure](#), [HTDL FQT Module Test Procedure](#)

**Conclusion/Witness/Outbrief**  
 None  
[Lab Range Outbrief](#)

Businesses because the testing was conducted at the Firmware level before the contractor has any low level firmware testing to verify SPS requirements.

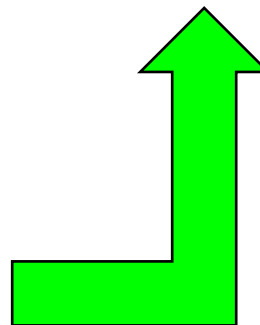
**Attachments**  
 (a) Lab Test Report, HTDL FQT Module Test Report, Module Credit Analysis  
 (4.11.11 Update), Rockwell C009-07B, CRIIS CFW STR\_946-4267-001 Rev A Draft 1.

**Conclusions:**  
 Government and contractor engineers discussed this verification at the PCA#1 planning meeting 3.2. The meeting leaders accepted the limited testing contained in Test Case WF-19 as verification on SP HTDL 66 when taking the following factors into consideration:

- A test of the ability to detect a deliberately corrupted message as erroneous passed.
- A test of the ability to pass a message that was not deliberately corrupted without declaring it erroneous passed.
- The SPS HTDL 66 does not specify the reliability or confidence of the error detection.
- The design uses a well-established technique for error detection so that the verification is test that the technique was implemented correctly in the HT-DLT and not testing the underlying ability of the technique.
- The design of the CRIIS system contains other separate message integrity checking at a high level than just the data link.
- Undetected message errors have not been problematic for the many hours of data link tested to date.

Signature: \_\_\_\_\_ Date: \_\_\_\_\_  
 Alan Masling 22 April 2015

Signature: \_\_\_\_\_ Date: \_\_\_\_\_  
 Kent A. Pouton 11 April 2015



Meet 2-3 times a Week w/ Ktr



# ***Functional Configuration Audit Results***

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- Incremental FCA approach allowed for earliest possible flight testing
- FCA-1 28-29 Apr 2015
  - 125 SPS Requirements accepted as complete after variances and actions closed
- FCA-2 on 26-27 Aug 2015
  - 20 SPS Requirements accepted as complete after variances and actions closed
- FCA-3 on 27-28 Jan 2016
  - 69 SPS Requirements accepted as complete after variances and actions closed

***214 SPS REQTS APPROVED AS PRODUCT BASELINE***

# Flight Test Overall Summary



# ***Flight Test Summary***

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- **Flight Test and SIL Execution Summary**
  - **21 Total Flight Test Events**
  - **F-15, F-16, F/A-18, L-29 and 4 Low Dynamic Aircraft**
  - **CRIIS System Successfully Integrated into Pax River and Eglin Range Infrastructure**
  - **SIL Completed for F-15, F-16, F/A-18, F-22 and F-35**
  
- **Flight Test Results Summary**
  - ✓ **All TSPI Performance Requirements Successfully Verified**
  - ✓ **All Datalink Networking and Reliability Requirements Successfully Verified**
  - ✓ **On-Aircraft MILS Demonstration Conducted Successfully**
  - ✓ **Aircraft Interface Requirements Successfully Verified at SIL**
  - ✓ **All Planned Flight Clearances Obtained and Airworthiness Data Completed**

# Eglin Flight Testing

	Aircraft	CRIIS Equipment	Objective	Execution Date(s)
<b>Eglin 1-4</b>	F-15	TMAC (Cfg5, Cfg6) Cfg4 x 2	TMAC TSPI Performance + Cfg4 TSPI Characterization	N/A
<b>Eglin L29</b>	L-29	<b>TMAC (Cfg5, Cfg6)</b> <b>Cfg4 x 1</b>	TMAC TSPI Performance + Cfg4 TSPI Characterization	10 Aug – 04 Sep 2015
<b>Eglin 5</b>	F-16	Cfg4 x 2	Cfg4 TSPI Characterization	20 Oct 2015
<b>MILS Demo</b>	F-15	Cfg4 x 1	MILS Demonstration	30 Sep 2015
<b>Regression</b>	As Needed	As Needed		N/A – No Regression Necessary

## ■ Eglin Changes from Plan

- F-15 Unavailable and Replaced 4 F-15 Flights with 13 L-29 Flights for TSPI Verification
  - Alternate F-15 Not Available due to Extensive Mod Required for TMAC Installation
  - L-29 Evaluated Against Planned Flight Cards to Ensure Representative High Dynamic Performance
- No Regression Test Required

# Eglin Flight Test (TSPI Verification)



- Two airborne systems on-board
- Scored against TSPI “Truth Plate”
- 13 test flights
- 197 total maneuvers



**KPP (0.5 meter Real Time horizontal RMS)**



**Real Time TSPI (positional & non-positional)**



**Post Mission Processed TSPI**

Maneuver	#
2-6 G turns	46
3-5 sec level inverted flight	31
Aileron rolls	28
Steep climbing or diving	24
Immelmann	14
2-4 G Figure-8	11
Cuban 8	10
Descending turns	10
Barrel rolls	9
Split S	8
High speed straight and level	6
Total	197

# Patuxent River NAS Flight Tests

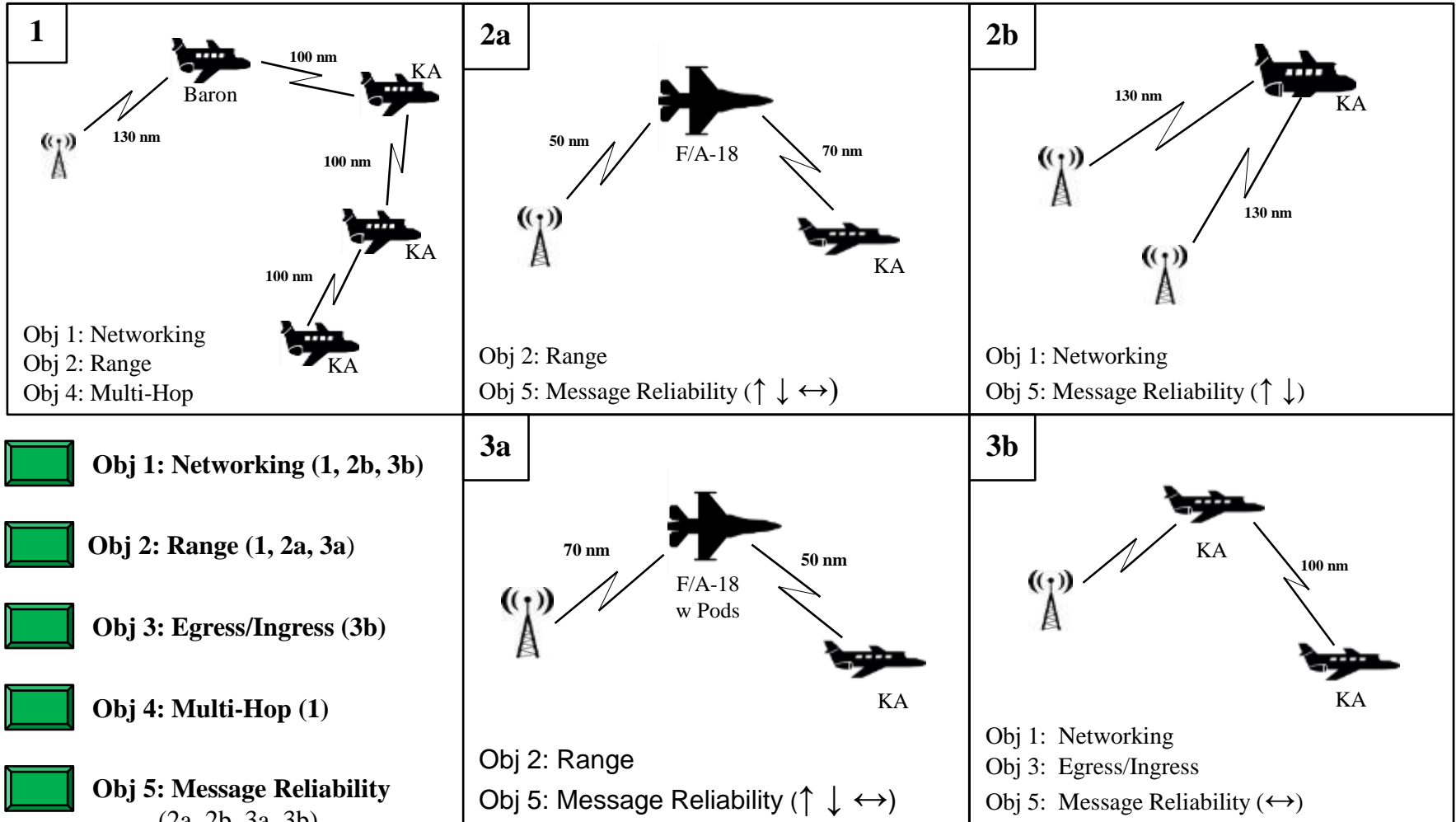
	Aircraft	CRIIS Equipment	Objective	Execution Date(s)
<b>Pax 1</b>	1 Baron, 3 x King Air	Cfg6 x 4	Multi-Hop, Datalink Routing & Message Reliability	16 July 2015
<b>Pax 2A</b>	F-18 1 x King Air	Cfg6 x 2	Datalink Distance & Message Reliability (2 RGS)	23 Sep 2015
<b>Pax 2B</b>	1 x King Air	Cfg6 x 1	Datalink Distance & Message Reliability	9 Sep 2015 22 Sep 2015 16 Oct 2015
<b>Pax 3A</b>	F-18 1 x King Air	Cfg4 x 2 Cfg6 x 1	Datalink Distance & Message Reliability (1 RGS)	30 Oct 2015
<b>Pax 3B</b>	2 x King Air	Cfg6 x 2	Datalink Distance & Message Reliability (1 RGS)	11 Sep 2015
<b>Regression</b>	1 x King Air	Cfg6 x 1	Message Reliability Confidence Flight (1&2 RGS)	03 Feb 2016

## ■ Pax River Changes from Plan

- Pax 2 and Pax 3 Separated into Two Parts to Ease Aircraft Scheduling Issues
- Additional Confidence Flight Added for Message Reliability Added



# Pax River Flight Test (Datalink Verification)



Meet with Build 33 software (2b Refly)

# ***Phased PCA Approach***

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- **Phase I: GSS/PTS HW (16-18 Nov 2015)**
  - FCA-1 and FCA-2 Completed Prior to Event
  - HW Only...No HW changes to GSS/PTS Expected
- **Phase II: C4/C5/C6 HW (20-21 Jan 2016)**
  - FCA-1, FCA-2 and Regression Testing Completed Prior to Event
  - Identified 4 Potential Variances from FCA
- **Phase III: System SW/Final HW (2-4 Feb 2016)**
  - FCA-1, FCA-2 and FCA-3 Completed Prior to Event

# *Results*

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- **Reviewed 327 HW & SW Parts with No Critical Issues Found**
  - Minor Configuration and Documentation Updates
- **Confirmed All Subsystems Consistent with Production Documentation**
- **Confirmed all Subsystems Functionally Equivalent with FCA Config**
  - Includes FCA Known Potential Variances
- **Completed Audit of SW and HW for All Subsystems**
  - Changes to these Parts Requires CCB
- **Verified SW Load Procedures for All Field Loadable CPINs**

***Successful PCA!***

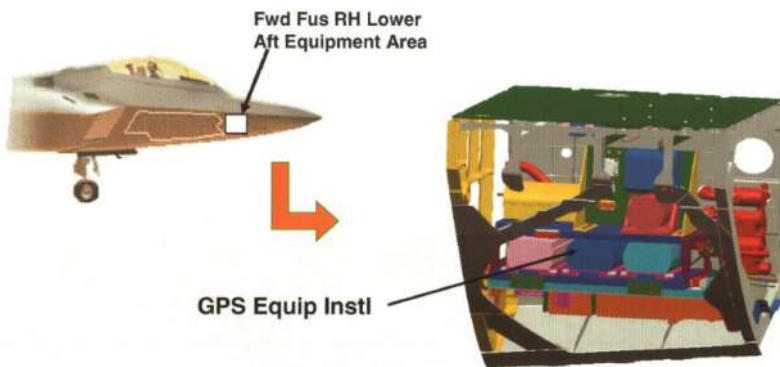
# ***CRIIS Upgrade/Integration***

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- **Lot 1 Production Contract Awarded 20 May 2016**
- **IDIQ - CRIIS software upgrade for advanced capabilities and to further mature integration with range infrastructure and aircraft bus messaging**
- **Initial Scope (Phase 1): 12 month effort to field CRIIS software upgrade**
  - **Modify CRIIS software to enable aircraft/range message handling**
  - **Datalink Power Switching for 5th Gen aircraft**
- **Follow-on efforts : F-15, F-16, F/A-18, F-22, F-35 integration verification**

# F-22-CRIIS Integration

- **Configuration 5 (2-Box) Targeted for 5th-Gen Aircraft**
  - Pre-CRIIS RFP analysis ID'd Air to Air Range Instrumentation (AARI) location
  - Operational Reqts superseded the available space forcing instrumentation relocation (ARIP)
    - Identified as the preferred location for CRIIS
  - Two CRIIS Config 5 installed as is in select A/C
- **CRIIS and F-22 Program MOA established to integrate onto F-22**
- **F-22 ID'd emerging ARDS obsolescence and spares issues**



ARIP →



# *Summary*

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- **CRIIS is next generation test range instrumentation system**
  - 5x data throughput
  - 4-20x TSPI accuracy
  - Legacy and 5th Gen platforms
- **CRIIS completed successful EMD; production deliveries started**
- **Follow-on efforts ongoing to continue to grow capability**